

Assume :

Newtonian, incompressible
steady-state

Since $L \gg R_2 - R_1$, assume flow
in gap is fully-developed.

For ease of calculation, I will fix my
local coord. syst. on falling cylinder.

$$\Rightarrow \text{Assume } v_x = v_x(y) \text{ only}$$

$$v_y = 0, \quad v_z = 0$$

Continuity

$$\frac{\partial v_x}{\partial x} + \frac{\partial v_y}{\partial y} + \frac{\partial v_z}{\partial z} = 0$$

$$0 = 0 \quad \checkmark \quad \text{satisfied}$$

x-comp of N-S.

$$\rho \left(\frac{\partial v_x}{\partial t} + v_x \frac{\partial v_x}{\partial x} + v_y \frac{\partial v_x}{\partial y} + v_z \frac{\partial v_x}{\partial z} \right) = -\frac{\partial p}{\partial x} + \eta \left(\frac{\partial^2 v_x}{\partial x^2} + \frac{\partial^2 v_x}{\partial y^2} + \frac{\partial^2 v_x}{\partial z^2} \right)$$

$$0 = -\frac{\partial p}{\partial x} + \eta \left(\frac{\partial^2 v_x}{\partial y^2} \right)$$

$$0 = -\frac{\partial p}{\partial x} + \eta \frac{d^2 v_x}{dy^2}$$

ordinary deriv., since
 $v_x = v_x(y)$ only

y-comp of N-S

$$0 = -\frac{\partial p}{\partial y}$$

$$\Rightarrow P = P(y)$$

z-comp of N-S

$$0 = -\frac{\partial p}{\partial z}$$

$$\Rightarrow P = P(z)$$

$$\Rightarrow \rho = \rho(x) \text{ only}$$

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so x -comp. becomes

$$0 = -\frac{d\rho}{dx} + \gamma \frac{d^2 v_x}{dy^2}$$

$$\underbrace{\frac{d\rho}{dx}}_{\text{function of } x \text{ only}} = \gamma \underbrace{\frac{d^2 v_x}{dy^2}}_{\text{function of } y \text{ only}} = \text{constant} = \frac{\Delta\rho}{L}$$

$$\frac{d^2 v_x}{dy^2} = \frac{1}{\gamma} \frac{\Delta\rho}{L}$$

$$\frac{dv_x}{dy} = \frac{1}{\gamma} \frac{\Delta\rho}{L} y + C_1$$

$$v_x = \frac{1}{2\gamma} \frac{\Delta\rho}{L} y^2 + C_1 y + C_2$$

In our coord. system, BC's are:

$$v_x(y=0) = 0$$

$$v_x(y=H) = U$$

substituting in:

$$0 = 0 + 0 + C_2 \Rightarrow C_2 = 0$$

$$U = \frac{1}{2\gamma} \frac{\Delta\rho}{L} H^2 + C_1 H$$

$$C_1 = \frac{U}{H} - \frac{1}{2\gamma} \frac{\Delta\rho}{L} H$$

$$v_x(y) = \frac{1}{2\gamma} \frac{\Delta\rho}{L} y^2 + \left(\frac{U}{H} - \frac{1}{2\gamma} \frac{\Delta\rho}{L} H \right) y$$

$$v_x(y) = U \left(\frac{y}{H} \right) - \frac{H^2}{2\gamma} \left(\frac{\Delta\rho}{L} \right) \left[\left(\frac{y}{H} \right) - \left(\frac{y}{H} \right)^2 \right]$$

Use conservation of mass, as in the gear pump problem, to solve for $\frac{\Delta P}{L}$.

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Here, the flow rate through the gap H must equal the rate of volume displacement by the cylinder.

That is, the fluid the solid cylinder displaces as it falls must flow up thru the gap.

$$\begin{aligned} Q &= \pi R_1^2 U = 2\pi R_1 \int_0^H v_x dy \\ &= 2\pi R_1 \int_0^H \left[U \left(\frac{y}{H} \right) - \frac{H^2}{2\gamma} \left(\frac{\Delta P}{L} \right) \left[\frac{y}{H} - \frac{y^2}{H^2} \right] \right] dy \\ &= 2\pi R_1 \left[\frac{U}{H} \frac{1}{2} y^2 \Big|_0^H - \frac{H}{2\gamma} \left(\frac{\Delta P}{L} \right) \frac{1}{2} y^2 \Big|_0^H + \frac{1}{2\gamma} \left(\frac{\Delta P}{L} \right) \frac{1}{3} y^3 \Big|_0^H \right] \end{aligned}$$

$$\pi R_1^2 U = 2\pi R_1 \left[\frac{1}{2} UH - \frac{1}{4} \frac{1}{\gamma} \left(\frac{\Delta P}{L} \right) H^3 + \frac{1}{6} \frac{1}{\gamma} \left(\frac{\Delta P}{L} \right) H^3 \right]$$

$$\pi R_1^2 U = 2\pi R_1 \left[\frac{UH}{2} - \frac{H^3}{12\gamma} \left(\frac{\Delta P}{L} \right) \right]$$

$$UR_1 - UH = -\frac{H^3}{6\gamma} \left(\frac{\Delta P}{L} \right)$$

$$UR_1 \left[1 - \frac{H}{R_1} \right] = -\frac{H^3}{6\gamma} \left(\frac{\Delta P}{L} \right)$$

$$\frac{\Delta P}{L} = -\frac{6\gamma R_1 U}{H^3} \left[1 - \frac{H}{R_1} \right] \approx -\frac{6\gamma R_1 U}{H^3}$$

may be neglected since $\frac{H}{R_1} \ll 1$

$$\Rightarrow v_x(y) = U \left(\frac{y}{H} \right) - \frac{H^2}{2\gamma} \left(-\frac{6\gamma R_1 U}{H^3} \right) \left[1 - \frac{H}{R_1} \right] \left(\frac{y}{H} - \frac{y^2}{H^2} \right)$$

in moving coord. system.

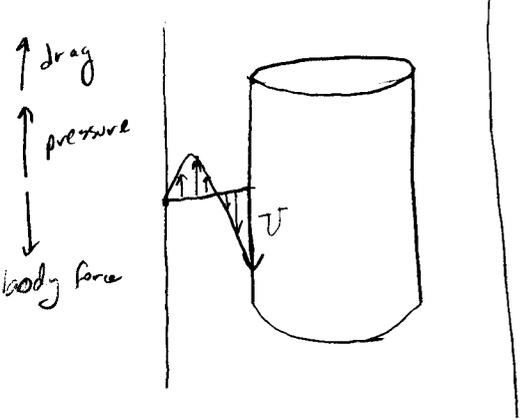
To get velocity in fixed coord system, just add $-U$ to velocity field, i.e.

$$v_x(y) = -U \left[1 - \frac{y}{H} \right] - \frac{\mu^2}{2\eta} \left(\frac{-6\eta R_1 U}{H^3} \right) \left[1 - \frac{H}{R_1} \right] \left(\frac{y}{H} - \frac{y^2}{H^2} \right)$$

(now, at $y=0$, $v_x = -U$
 at $y=H$, $v_x = 0$)

Now we write a force balance on the cylinder to relate η to U .

We are interested in the x-comp of the force



The forces acting on cylinder are a body force, a drag force due to T_{yx} , and a pressure force ~~due~~ acting on ends of cylinder due to ΔP (note this includes gravitational force).

$$\sum F_x = 0 = \underbrace{-\pi R_1^2 L (\rho_c - \rho) g}_{\substack{\text{volume} \\ \text{of fluid} \\ \text{displaced}}} + \underbrace{\pi R_1^2 [P(0) - P(L)]}_{\substack{\text{body force} \\ \text{includes} \\ \text{buoyancy}}} + \underbrace{2\pi R_1 L T_{yx} \Big|_{y=0}}_{\text{drag force}}$$

difference in pressures acting on two ends of cylinders

From part (a), we know $\frac{dP}{dx} = \frac{\Delta P}{L} = \frac{-6\eta R_1 U}{H^3} \left[1 - \frac{H}{R_1} \right]$

$$\Rightarrow P(x) = -\frac{6\eta R_1 U}{H^3} \left[1 - \frac{H}{R_1} \right] x + C_0$$

at $x=0$, $P=P_0 \Rightarrow C_0 = P_0 = P(0)$

$$\Rightarrow P(L) = -\frac{6\eta R_1 U}{H^3} \left[1 - \frac{H}{R_1} \right] L + P(0)$$

$$\Rightarrow P(0) - P(L) = \frac{6\eta R_1 U}{H^3} \left[1 - \frac{H}{R_1} \right] L \approx \frac{6\eta R_1 U L}{H^3}$$

We can evaluate $\tau_{yx} \Big|_{y=0}$ from the velocity field:

$$\begin{aligned} \tau_{yx} \Big|_{y=0} &= \eta \frac{dv_x}{dy} \Big|_{y=0} = \eta \left[\frac{U}{H} - \frac{H^2}{2\eta} \left(-\frac{6\eta R_1 U}{H^3} \right) \left(1 - \frac{H}{R_1} \right) \left(\frac{1}{H} - \frac{2y}{H^2} \right) \right] \Big|_{y=0} \\ &= \eta \left[\frac{U}{H} + \frac{3R_1 U}{H} \left(1 - \frac{H}{R_1} \right) \left(\frac{1}{H} \right) \right] \\ &= \frac{\eta U}{H} \left[1 + \frac{3R_1}{H} \left(1 - \frac{H}{R_1} \right) \right] \\ &\approx \frac{\eta U}{H} \left[1 + \frac{3R_1}{H} \right] \approx \frac{\eta U}{H} \left(\frac{3R_1}{H} \right) \approx \frac{3\eta U R_1}{H^2} \end{aligned}$$

The drag force is

$$F_{\text{drag}} = \tau_{yx} \Big|_{y=0} A = 2\pi R_1 L \left(\frac{3\eta U R_1}{H^2} \right) \leftarrow \text{answer to (b)}$$

Substituting into our force balance:

$$0 = -\pi R_1^2 L (p_c - p) g + \pi R_1^2 \left(\frac{6\eta R_1 U L}{H^3} \right) + 2\pi R_1 L \left(\frac{3\eta U R_1}{H^2} \right)$$

Simplifying

$$0 = -(\rho_c - \rho)g + \frac{6\gamma R_1 U}{H^3} + \frac{6\gamma U}{H^2}$$

$$(\rho_c - \rho)g = \frac{6\gamma R_1 U}{H^3} \left(1 + \frac{H}{R_1} \right) \approx \frac{6\gamma R_1 U}{H^3}$$

$$\Rightarrow \boxed{\eta = \frac{(\rho_c - \rho)g H^3}{6R_1 U}}$$

2. a). Assume Newtonian, incompressible fluid, steady, $Re \ll 1$ (creeping flow)

Because of symmetry, assume v and ρ are not functions of ϕ

Assume, $v_r = 0, v_\theta = 0 \quad \& \quad v_\phi = v_\phi(r, \theta)$

$\rho = \rho(r, \theta)$ by symmetry

Continuity:

$$0 = \frac{1}{r^2} \frac{\partial}{\partial r} (r^2 v_r) + \frac{1}{r \sin \theta} \frac{\partial}{\partial \theta} (v_\theta \sin \theta) + \frac{1}{r \sin \theta} \frac{\partial}{\partial \phi} (v_\phi)$$

$$0 = 0$$

b) r-comp of N-S

$$0 = -\frac{\partial \rho}{\partial r} \Rightarrow \rho \neq \rho(r)$$

θ -comp of N-S

$$0 = -\frac{1}{r} \frac{\partial \rho}{\partial \theta} \Rightarrow \rho \neq \rho(\theta)$$

} $\Rightarrow \underline{\rho = \text{constant!}}$

ϕ -comp of N-S

$$0 = \left[\frac{1}{r^2} \frac{\partial}{\partial r} \left(r^2 \frac{\partial v_\phi}{\partial r} \right) + \frac{1}{r^2 \sin \theta} \frac{\partial}{\partial \theta} \left(\sin \theta \frac{\partial v_\phi}{\partial \theta} \right) + \frac{1}{r^2 \sin^2 \theta} \frac{\partial^2 v_\phi}{\partial \phi^2} - \frac{v_\phi}{r^2 \sin^2 \theta} \right]$$

$$0 = \frac{\partial}{\partial r} \left(r^2 \frac{\partial v_\phi}{\partial r} \right) + \frac{1}{\sin \theta} \frac{\partial}{\partial \theta} \left(\sin \theta \frac{\partial v_\phi}{\partial \theta} \right) - \frac{v_\phi}{\sin^2 \theta}$$

BC's : $v_\phi = 0$ at $r = kR$
 $v_\phi = \Omega R \sin \theta$ at $r = R$ (no slip)

c) As in other creeping flow problems we solved, let's guess that the solution retains the same dependence on θ throughout the domain.

That is, let's try a solution of the form

$$v_{\theta} = f(r) \sin \theta$$

that satisfies BC's $f(r=kR) = 0$

$$f(r=R) = \Omega R$$

Substitute this into our circled equation above:

$$0 = \frac{\partial}{\partial r} \left(r^2 \frac{\partial}{\partial r} (f \sin \theta) \right) + \frac{1}{\sin \theta} \frac{\partial}{\partial \theta} \left(\sin \theta \frac{\partial}{\partial \theta} (f \sin \theta) \right) - \frac{f \sin \theta}{r^2}$$

$$0 = \sin \theta \frac{d}{dr} \left(r^2 \frac{df}{dr} \right) + \frac{f}{\sin \theta} \frac{d}{d\theta} (\sin \theta \cos \theta) - \frac{f}{r^2}$$

$$0 = \sin \theta \frac{d}{dr} \left(r^2 \frac{df}{dr} \right) + \frac{f}{\sin \theta} \left(-\sin^2 \theta + \underbrace{\cos^2 \theta}_{1 - \sin^2 \theta} \right) - \frac{f}{r^2}$$

$$0 = \sin \theta \frac{d}{dr} \left(r^2 \frac{df}{dr} \right) + \frac{f}{\sin \theta} \left(\cancel{1} - 2 \sin^2 \theta \right) - \frac{f}{r^2}$$

$$0 = \frac{d}{dr} \left(r^2 \frac{df}{dr} \right) - 2f$$

$$\text{BCs } f(kR) = 0$$

$$f(R) = \Omega R$$

Expanding:

$$r^2 \frac{d^2 f}{dr^2} + 2r \frac{df}{dr} - 2f = 0$$

Using hint provided: $a=1$ $b=2$ $c=-2$

$$n(n-1) + 2n - 2 = 0$$

$$n^2 - n + 2n - 2 = 0$$

$$n^2 + n - 2 = 0$$

$$(n+2)(n-1) = 0 \Rightarrow n = -2, n = 1$$

$$\text{Solution is } f = C_1 r^{-2} + C_2 r^1$$

Applying BC at $r=kR$:

$$0 = c_1 \frac{1}{(kR)^2} + c_2 (kR)$$

$$c_1 = -c_2 (kR)^3$$

Applying BC at $r=R$

$$\Omega R = -c_2 (kR)^3 \frac{1}{R^2} + c_2 R = -c_2 (k^3 R - R)$$

$$\Omega R = c_2 R (1 - k^3)$$

$$\Rightarrow c_2 = \frac{\Omega}{1 - k^3}$$

$$\Rightarrow c_1 = -\frac{\Omega}{1 - k^3} (kR)^3$$

$$f = -\frac{\Omega}{1 - k^3} (kR)^3 \frac{1}{r^2} + \frac{\Omega}{1 - k^3} r$$

$$f = \frac{\Omega}{1 - k^3} \left(r - \frac{(kR)^3}{r^2} \right)$$

$$\Rightarrow \boxed{V_\phi = \frac{\Omega}{1 - k^3} \left(r - \frac{(kR)^3}{r^2} \right) \sin \theta}$$

d). largest inertial term discarded from r-comp is:

$$\frac{\rho V_\phi^2}{r} \sim \frac{\rho}{R} \left(\frac{\Omega R}{1 - k^3} \right)^2 \sim \frac{\rho \Omega^2 R}{(1 - k^3)^2}$$

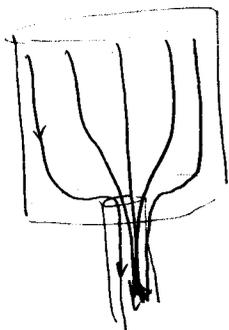
Viscous term is

$$\eta \frac{1}{r^2} \frac{\partial}{\partial r} \left(r^2 \frac{\partial v_\phi}{\partial r} \right) \sim \eta \frac{1}{R^2} \frac{1}{R} \left(R^2 \frac{1}{R} \frac{\Omega}{1 - k^3} R \right)$$

$$\text{viscous} \sim \gamma \frac{1}{R} \frac{\Omega}{1-k^3}$$

$$\text{Ratio of } \frac{\text{inertial}}{\text{viscous}} \sim \frac{\rho \Omega^2 R}{(1-k^3)^2} \frac{R(1-k^3)}{\gamma \Omega} \sim \frac{\rho \Omega R^2}{\gamma (1-k^3)}$$

We are interested in flow near the contraction plane, where streamlines will likely look like:



So we'll assume:

Newtonian
Incompressible
Steady

H is comparable to R_1

Axisymmetric \Rightarrow no θ dependence

Assume

a)

$$\begin{aligned} v_\theta &= 0 \\ v_r &= v_r(r, z) \\ v_z &= v_z(r, z) \end{aligned}$$

$$p = p(r, z)$$

$$b) \quad v_r = 0 \quad \begin{array}{l} \text{at } r = R_1 \quad \text{for } z < 0 \\ \text{at } r = R_2 \quad \text{for } z > 0 \end{array}$$

$$v_r = 0 \quad \text{at } r = 0 \quad \text{for all } z$$

$$v_z = 0 \quad \begin{array}{l} \text{at } r = R_1 \quad \text{for } z < 0 \\ \text{at } r = R_2 \quad \text{for } z > 0 \end{array}$$

$$v_z = \text{finite at } r = 0 \quad \text{OR} \quad \frac{\partial v_z}{\partial r} = 0 \quad \text{at } r = 0$$

$$\text{Note: } \frac{\partial v_r}{\partial r} \Big|_{r=0} \neq 0 \quad \text{since} \quad \frac{\partial v_z}{\partial z} \Big|_{r=0} \neq 0 \quad (\text{see continuity}).$$

c) r-comp: becomes:

$$\rho \left(v_r \frac{\partial v_r}{\partial r} + v_z \frac{\partial v_r}{\partial z} \right) = -\frac{\partial p}{\partial r} + \eta \left[\frac{\partial}{\partial r} \left(\frac{1}{r} \frac{\partial}{\partial r} (r v_r) \right) + \frac{\partial^2 v_r}{\partial z^2} \right]$$

θ -comp becomes:

$$0 = -\frac{1}{r} \frac{\partial p}{\partial \theta} \Rightarrow p \neq p(\theta)$$

z-comp becomes

3-2

$$\rho \left(r_r \frac{\partial v_z}{\partial r} + r_z \frac{\partial v_z}{\partial z} \right) = -\frac{\partial p}{\partial z} + \eta \left[\frac{1}{r} \frac{\partial}{\partial r} \left(r \frac{\partial v_z}{\partial r} \right) + \frac{\partial^2 v_z}{\partial z^2} \right]$$

d). From our boundary layer approx, we got

$$\frac{L_e}{D} \approx 0.01 Re$$

$$z \approx 0.01 \left(\frac{\rho \langle v \rangle D^2}{\eta} \right) = 0.01 \left(\frac{4 \rho \langle v \rangle R_2^2}{\eta} \right)$$

Basing your answer on Eqn 3.14a is ok too.